

SSLS111 - Simple eccentricing of plate

Summarized:

This test makes it possible to validate the eccentricing of the simple plates (i.e. that it is neither about multi-layer, nor of a homogenized behavior).

Three modelizations are used: DST (modelization *A*), DKT (modelization *B*) and DST (modelization *C*).

Modelizations *A* and *B* : one models 2 plates offset compared to the average plane and one compares the results with an analytical solution.

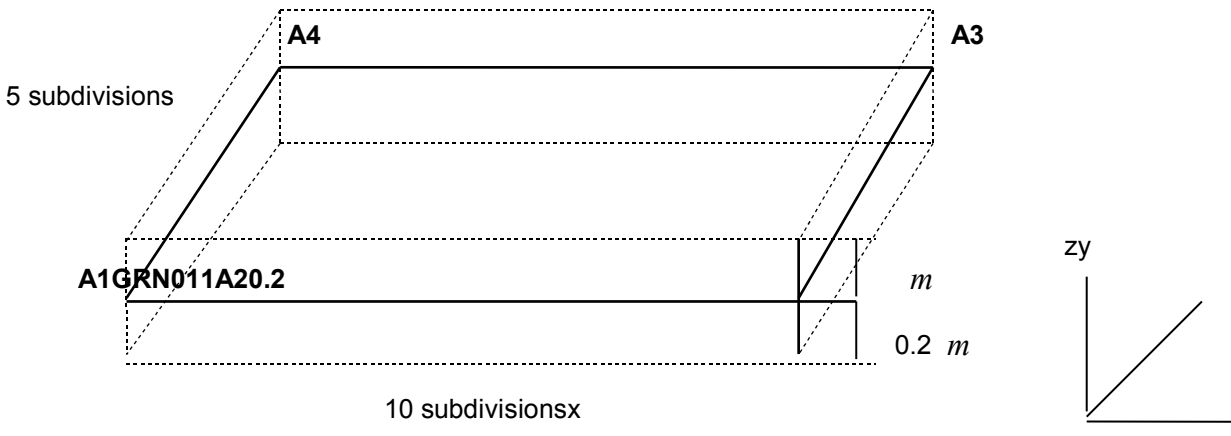
Modelization *C* :

The reference is given by a first resolution where one models double-layered made up of 2 materials.

The validation is done in the second computation where one models the 2 layers of the preceding model by 2 plates offset compared to the average plane of the first computation.

1 Problem of reference

1.1 Geometry



the coordinates of the points are given in meters (m):

$$\begin{array}{ll} A1(0,0,0) & A3(10,5,0) \\ A2(10,0,0) & A4(0,5,0) \end{array}$$

1.2 Material properties

1.2.1 Modelizations A and B

the material has an isotropic elastic behavior:

Young modulus: $E = 20000.MPa$

Poisson's ratio: $\nu = 0.$

Density: $\rho = 1000.Kg/m^3$

1.2.2 Modelization C

the material is double-layered.

The material constituting the first layer is elastic orthotropic and is characterized by the following data:

$$EL = 20000.MPa \quad ET = 20000.MPa \quad VLT = 0.3 \quad GLT = 2000.MPa.$$

The material constituting the second layer is also elastic orthotropic and is characterized by the following data:

$$EL = 15000.MPa \quad ET = 15000.MPa \quad VLT = 0.3 \quad GLT = 1500.MPa$$

1.3 Boundary conditions and loadings

1.3.1 Modelizations A and B

the edge $A1A4$ is clamped.
One applies a nodal force $F_z = -1000$ N to the edge $A1A2$.

1.3.2 Modelization C

the node $A1$ is clamped:

$$\begin{array}{lll} dx=0. & dy=0. & dz=0. \\ dRx=0. & dRy=0. & dRz=0. \end{array}$$

The node $A2$ is blocked according to the following degrees of freedom:

$$dx=0. \quad dy=0.$$

One applies a nodal force $F_z = -1000$ N to the node $A3$.

In addition, one applies to 3 meshes (see drawing into 5.2) the distributed loading in local coordinate system according to (key word `FORCE_COQUE`):

$$F1=200 \text{ N/m}^2 \quad F2=-500 \text{ N/m}^2 \quad F3=-500 \text{ N/m}^2 \quad MF1=100 \text{ N/m} , \quad MF2=40 \text{ N/m}$$

in the plane of the mesh.

2 Reference solution

2.1 Méthode de calcul used for the reference solution

2.1.1 Modelizations A and B

the deflection f_l is given by the formula: $f_l = FIL^3/3EI$
where l is the width, L the length of the plate, and $I = lh^3/12$, h being the thickness.

2.1.2 Modelization C

The computation with double-layered material is used as reference. The non regression one of the results got for this first computation is checked.

2.2 Results of reference

2.2.1 Modelizations A and B

They are made up by the values of the field of displacement DZ at the point $A3$ and the forces at the point AI . In addition, the 4 smaller frequencies of structure are calculated.

2.2.2 Modelization C

They are made up by the values of the field of displacement DX, DY, DZ, DRX, DRY at the point $A3$ (node NI for Code_Aster) and at the point of coordinates $(9,2,0)$.

One compares also the forces with the point AI .

In addition, the 4 smaller frequencies of structure are calculated.

2.3 Uncertainty on the solution

For the modelizations A and B , the reference solution is analytical. There is thus no uncertainty.

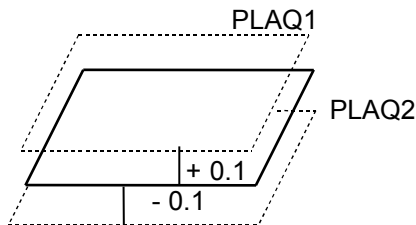
For the modelization C , uncertainties are null since it is about the same computation carried out by two different ways.

3 Modelization A

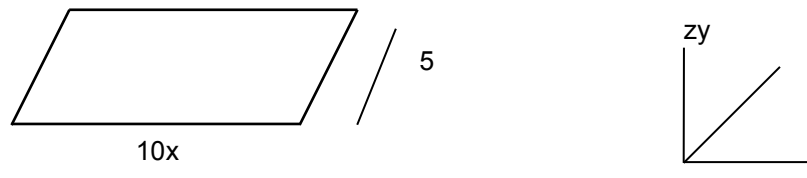
3.1 Characteristic of the modelization

The model consists of 2 offset plates of the distances -0.1m and 0.1m .

The elements used are shell elements DSQ.



3.2 Characteristics of the mesh



The mesh is regular. There are 10 subdivisions according to x and 5 subdivisions according to y ; that is to say on the whole 50 meshes DSQ (QUAD4) and 66 nodes.

3.3 Quantities tested and Standard

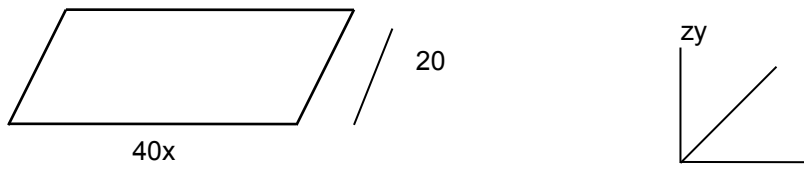
Identification	results of reference	Values of reference	Tolerance
DZ (A3)	"ANALYTIQUE"	$-3.90625 \cdot 10^{-5}$	0.005
MXX (A1)	"ANALYTIQUE"	5000.	0.001
QX (A1)	"ANALYTIQUE"	-500.	0.001
SIXX (MI , PT1 , SSPT1)	"ANALYTIQUE"	0.	1.0E-8
SIXX (MI , PT1 , SSPT2)	"ANALYTIQUE"	990.585	0.001
Frequency 1st mode	"NON_REGRESSION"	18.2	0.001
Frequency 2nd mode	"NON_REGRESSION"	84.6	0.001
Frequency 3rd mode	"NON_REGRESSION"	101.21	0.001
Frequency 4th mode	"NON_REGRESSION"	111.28	0.001

4 Modelization B

4.1 Characteristic of the modelization

The model is the same one as that of the modelization A, with this close instead of having shell elements `DSQ`, one has elements `DKT`.

4.2 Characteristic of the mesh



The mesh is regular. There are 40 subdivisions according to x and 20 subdivisions according to y ; i.e. a total of 1600 meshes `DKT` and 861 nodes.

4.3 Quantities tested and Standard

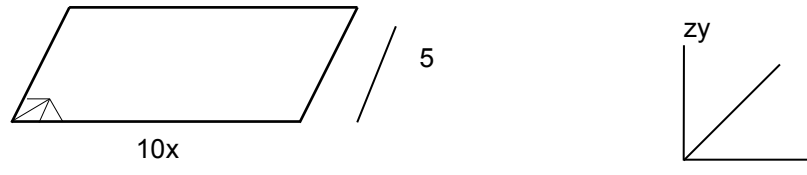
Identification	results of reference	Values of reference	Tolerance
DZ (A3)	"ANALYTIQUE"	- 3.90625 10-5	0.005
MXX (A1)	"ANALYTIQUE"	5000.	0.3
QX (A1)	"ANALYTIQUE"	-500.	0.15
Frequency 1st mode	"NON_REGRESSION"	18.25	0.001
Frequency 2nd mode	"NON_REGRESSION"	88.35	0.001
Frequency 3rd mode	"NON_REGRESSION"	100.1	0.001
Frequency 4th mode	"NON_REGRESSION"	113.5	0.001

5 Modelization C

5.1 Characteristic of the modelization

The model are the same one as that of the modelization *A* , with this close instead of having shell elements *DSQ* , one has elements *DST* . (Modelization *DST* with meshes *TRIA3*).

5.2 Characteristics of the mesh



The mesh is regular. There are 10 subdivisions according to *x* and 5 subdivisions according to *y* ; i.e. a total of 100 meshes *DST* and 66 nodes.

5.3 Quantities tested and Standard

Identification	results of reference	Values of reference	Tolerance
DEPL DX <i>N66</i>	"AUTRE_ASTER"	- 6.49678E-06	0.01
DEPL DY <i>N66</i>	"AUTRE_ASTER"	- 6.08932E-07	0.01
DEPL DZ <i>N66</i>	"AUTRE_ASTER"	- 5.33844E-03	0.01
DEPL DRX <i>N66</i>	"AUTRE_ASTER"	- 4.29182E-04	0.01
DEPL DRY <i>N66</i>	"AUTRE_ASTER"	4.75601E-04	0.01
DEPL DX <i>N53</i>	"AUTRE_ASTER"	-3.58293E-06	0.01
DEPL DY <i>N53</i>	"AUTRE_ASTER"	- 1.18788E-06	0.01
DEPL DZ <i>N53</i>	"AUTRE_ASTER"	- 3.63885E-03	0.01
DEPL DRX <i>N53</i>	"AUTRE_ASTER"	- 4.05175E-04	0.01
DEPL DRY <i>N53</i>	"AUTRE_ASTER"	4.23116E-04	0.01
EFGE NXX <i>N6</i>	"AUTRE_ASTER"	1.70005E+04	0.01
EFGE NYY <i>N6</i>	"AUTRE_ASTER"	1.14438E+04	0.016
EFGE NXY <i>N6</i>	"AUTRE_ASTER"	3.53598E+03	0.022
EFGE MXX <i>N6</i>	"AUTRE_ASTER"	2.14585E+04	0.01
EFGE MYY <i>N6</i>	"AUTRE_ASTER"	1.53094E+04	0.012
EFGE MXY <i>N6</i>	"AUTRE_ASTER"	5.71331E+03	0.02
EFGE QX <i>N6</i>	"AUTRE_ASTER"	- 3.03380E+03	0.08
EFGE QY <i>N6</i>	"AUTRE_ASTER"	1.76436E+03	0.025
MODE 1	"AUTRE_ASTER"	1.01181E+00	0.003
MODE 2	"AUTRE_ASTER"	4.27003E+00	0.003
MODE 3	"AUTRE_ASTER"	8.39151E+00	0.004
MODE 4	"AUTRE_ASTER"	1.72305E+01	0.006

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5.4 Remarks

One notes a difference between the solution obtained for a double-layered shell and that resulting from two offset full-course shells, without it being possible at the time of the drafting of the test to determine from which the variation comes.

6 Summary of the results

the results of the 3 modelizations are very good, except for the shears of the modelizations *B* and *C* (respectively 10% and 7% of error).